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2 3 This invention relates to a method and apparatus for 4 moving a fluid. 5 6 The present invention has reference to improvements to a fluid mover having a number of practical .7 8 applications of diverse nature ranging from marine 9 propulsion systems to pumping applications for 10 moving and/or mixing fluids and/or solids of the same or different characteristics. The present 11 invention also has relevance in the fields inter 12 13 alia of heating, cooking, cleaning, aeration, gas 14 fluidisation, and agitation of fluids and 15 fluids/solids mixtures, particle separation, 16 classification, disintegration, mixing, 17 emulsification, homogenisation, dispersion, maceration, hydration, atomisation, droplet 18 19 production, viscosity reduction, dilution, shear 20 thinning, transport of thixotropic fluids and 21 pasteurisation.

METHOD AND APPARATUS FOR MOVING A FLUID

1 More particularly the invention is concerned with 2 the provision of an improved fluid mover having 3 essentially no moving parts. 4 Ejectors are well known in the art for moving 6 working or process fluids by the use of either a central or an annular jet which emits steam into a 8 duct in order to move the fluids through or out of 9 appropriate ducting or into or through another body 10 . of fluid. The ejector principally operates on the 11 12 basis of inducing flow by creating negative 13 pressure, generally by the use of the venturi 14 principle. The majority of these systems utilise a 15 central steam nozzle where the induced fluid generally enters the duct orthogonally to the axis 16 of the jet, although there are exceptions where the 17 reverse arrangement is provided. The steam jet is 18 accelerated through an expansion nozzle into a 19 20 mixing chamber where it impinges on and is mixed 21 with working fluid. The mixture of working fluid 22 and steam is accelerated to higher velocities within a downstream convergent section prior to a divergent 23 section, e.g. a venturi. The pressure gradient 24 generated in the venturi induces new working fluid 25 26 to enter the mixing chamber. The energy transfer 27 mechanism in most steam ejector systems is a combination of momentum, heat and mass transfer but 28 29 by varying proportions. Many of these systems 30 employ the momentum transfer associated with a 31 converging flow, while others involve the generation of a shock wave in the divergent section. One of 32

working fluid passing through the centre of the 2 hollow body. 3 PCT/GB2003/004400 describes that the transport fluid 4 5 is preferably a condensable fluid and may be a gas or vapour, for example steam, which may be 6 introduced in either a continuous or discontinuous 7 8 manner. At or near the point of introduction of the transport fluid, for example immediately downstream 10 thereof, a pseudo-vena contracta or pseudo 11 convergent/divergent section is generated, akin to the convergent/divergent section of conventional 12 13 steam ejectors but without the physical constraints associated therewith since the relevant section is 14 15 formed by the effect of the steam impacting upon the 16 working or process fluid. Accordingly the fluid mover is more versatile than conventional ejectors 17 18 by virtue of a flexible fluidic internal boundary 19 described by the pseudo-vena contracta. 20 flexible boundary lies between the working fluid at 21 the centre and the solid wall of the unit, and allows disturbances or pressure fluctuations in the 22 23 multi phase flow to be accommodated better than for 24 a solid wall. This advantageously reduces the 25 supersonic velocity within the multi phase flow, 26 resulting in better droplet dispersion, increasing 27 the momentum transfer zone length, thus producing a more intense condensation shock wave. 28 29 30 PCT/GB2003/004400 further discloses that the 31 positioning and intensity of the shock wave is variable and controllable depending upon the 32

1 specific requirements of the system in which the 2 fluid mover is disposed. The mechanism relies on a combination of effects in order to achieve its high 3 versatility and performance, notably heat, momentum 4 5 and mass transfer which gives rise to the generation of the shock wave and also provides for shearing of 6 7 the working fluid flow on a continuous basis by 8 shear dispersion and/or dissociation. Preferably the nozzle is located as close as possible to the 9 10 projected surface of the working fluid in practice 11 and in this respect a knife edge separation between 12 the transport fluid or steam and the working fluid 13 stream is of advantage in order to achieve the 14 requisite degree of interaction. The angular 15 orientation of the nozzle with respect to the 16 working fluid stream is of importance and may be 17 shallow. 18 Further, PCT/GB2003/004400 discloses that the or 19 20 each transport fluid nozzle may be of a convergent-21 divergent geometry internally thereof, and in 22 practice the nozzle is configured to give the 23 supersonic flow of transport fluid within the passage. For a given steam condition, i.e. dryness, 24 25 pressure and temperature, the nozzle is preferably 26 configured to provide the highest velocity steam 27 jet, the lowest total pressure drop and the highest 28 static enthalpy between the steam chamber and the 29 nozzle exit. The nozzle is preferably configured to 30 avoid any shock in the nozzle itself. For example 31 only, and not by way of limitation, an optimum area 32 ratio for the nozzle, namely exit area: throat area,

lies in the range 1.75 and 7.5, with an included 1 angle of less than 9°. 2 3 The or each nozzle is conveniently angled towards 4 the working fluid flow since this helps penetration 5 of the working fluid by the transport fluid, which 6 7 may help shear or thermal dispersion of the working fluid. This may also prevent both kinetic energy 8 9 dissipation on the wall of the passage and premature 10 condensation of the steam at the wall of the passage, where an adverse temperature differential 11 12 prevails. The angular orientation of the nozzles is 13 selected for optimum performance which is dependent 14 inter alia on the nozzle orientation and the internal geometry of the mixing chamber. Further 15 16 the angular orientation of the or each nozzle is selected to control the pseudo-convergent/divergent 17 · profile, the pressure profile within the mixing 18 19 chamber, the enthalpy addition and the condensation 20 shock wave intensity or position in accordance with 21 the pressure and flow rates required from the fluid 22 Moreover, the creation of turbulence, 23 governed inter alia by the angular orientation of 24 the nozzle, is important to achieve optimum 25 performance by dispersal of the working fluid to a 26 vapour-droplet phase in order to increase 27 acceleration by momentum transfer. This aspect is of 28 particular importance when the fluid mover is employed as a pump. For example, and not by way of 29 30 limitation, in the present invention it has been 31 found that an angular orientation for the or each

nozzle may lie in the range 0 to 30° with respect to 1 the flow direction of the working fluid. 2 3 A series of nozzles with respective mixing chamber 4 sections associated therewith may be provided 5 longitudinally of the passage and in this instance 6 the nozzles may have different angular orientations, 7 for example decreasing from the first nozzle in a 8 downstream direction. Each nozzle may have a 9 different function from the other or others, for 10 example pumping, mixing, disintegrating, and may be 11 selectively brought into operation in practice. 12 Each nozzle may be configured to give the desired 13 . effects upon the working fluid. Further, in a 14 multi-nozzle system by the introduction of the 15 transport fluid, for example steam, phased heating 16 may be achieved. This approach may be desirable to 17 provide a gradual heating of the working fluid. 18 19 An object of the present invention is to improve the 20 performance of the fluid mover by enhancing the 21 energy transfer mechanism between the high velocity 22 transport fluid and the working fluid. 23 improves the performance of the fluid mover having 24 essentially no moving parts having an improved 25 performance than fluid movers currently available in 26 the absence of any constriction such as is 27 exemplified in the prior art recited in the 28 aforementioned patent. 29 30 According to a first aspect of the present invention 31 a fluid mover includes a hollow body provided with a 32

straight-through passage of substantially constant 1 cross section with an inlet at one end of the 2 3 passage and an outlet at the other end of the 4 passage for the entry and discharge respectively of 5 a working fluid, a nozzle substantially circumscribing and opening into said passage 6 intermediate the inlet and outlet ends thereof, an 7 8 inlet communicating with the nozzle for the 9 introduction of a transport fluid, a mixing chamber 10 being formed within the passage downstream of the nozzle, the nozzle internal geometry and the bore 11 profile immediately upstream of the nozzle exit 12 13 being so disposed and configured to optimise the 14 energy transfer between the transport fluid and 15 working fluid that in use through the introduction 16 of transport fluid the working fluid or fluids are atomised to form a dispersed vapour/droplet flow 17 18 regime with locally supersonic flow conditions 19 within a pseudo-vena contracta, resulting in the 20 creation of a supersonic condensation shock wave 21 within the downstream mixing chamber by the condensation of the transport fluid. 22 23 24 The transport fluid is preferably a condensable 25 fluid and may be a gas or vapour, for example steam, 26 which may be introduced in either a continuous or 27 discontinuous manner. 28 29 According to a second aspect of the present 30 invention a fluid mover of the kind described in our 31 aforementioned patent application, includes a hollow 32 body provided with a straight-through passage of

substantially constant cross section with an inlet 1 at one end of the passage and an outlet at the other 2 end of the passage for the entry and discharge 3 respectively of a working fluid, a nozzle 4 substantially circumscribing and opening into said 5 passage intermediate the inlet and outlet ends 6 thereof, an inlet communicating with the nozzle for 7 the introduction of steam, a mixing chamber being 8 formed within the passage downstream of the nozzle, 9 the nozzle internal geometry and the bore profile 10 immediately upstream of the nozzle exit being so 11 disposed and configured to optimise the energy 12 transfer between the steam and working fluid that in 13 use through the introduction of steam the working 14 fluid or fluids are atomised to form a dispersed 15 vapour/droplet flow regime with locally supersonic 16 flow conditions within a pseudo-vena contracta, 17 resulting in the creation of a supersonic 18 condensation shock wave within the downstream mixing chamber by the condensation of the steam. 20 21 The nozzle may be of a form to correspond with the 22 shape of the passage and thus for example a circular 23 passage would advantageously be provided with an 24 annular nozzle circumscribing it. The term 25 'annular' as used herein is deemed to embrace any 26 configuration of nozzle or nozzles that 27 circumscribes the passage of the fluid mover, and 28 encompasses circular, irregular, polygonal and 29 rectilinear shapes of nozzle. 30

The or each nozzle may be of a convergent-divergent 1 2 geometry internally thereof, and in practice the 3 nozzle is configured to give the supersonic flow of transport fluid within the passage. For a given 4 5 steam condition, i.e. dryness, pressure and temperature, the nozzle is preferably configured to 6 7 provide the highest velocity steam jet, the lowest total pressure drop and the highest enthalpy between 8 the steam chamber and nozzle exit. 9 10 11 The condensation profile in the mixing chamber 12 determines the expansion ratio profile across the 13 nozzle. With relatively low working fluid 14 temperatures condensation is dominant, and the exit 15 pressure of the transport fluid nozzle is low. The 16 exit pressure of the transport fluid nozzle is 17 higher when the bulk temperature of the working 18 fluid is higher. 19 According to a third aspect of the present invention 20 21 a method of moving a working fluid includes 22 presenting a fluid mover to the working fluid, 23 the mover having a straight-through passage of 24 substantially constant cross section, 25 applying a substantially circumscribing stream 26 of a transport fluid to the passage through an 27 annular nozzle, 28 atomising the working fluid to form a dispersed vapour and droplet flow regime with locally 29 30 supersonic flow conditions,

generating a supersonic condensation shock wave 1 within the passage downstream of the nozzle by 2 condensation of the transport fluid, 3 inducing flow of the working fluid through the 4 passage from an inlet to an outlet thereof, and 5 modulating the condensation shock wave to vary 6 the working fluid discharge from the outlet. 7 8 Preferably the modulating step includes modulating 9 the intensity of the condensation shock wave. 10 Alternatively or additionally the modulating step 11 includes modulating the position of the condensation 12 shock wave. 13 14 The bore profile immediately upstream of the nozzle 15 is preferably configured to encourage working fluid 16 atomisation. Preferably an instability in working 17 fluid flow is introduced immediately upstream of the 18 19 20 The or each nozzle is preferably optimally 21 configured to operate with a particular working 22 fluid, upstream wall contour profile and mixing 23 chamber geometry. The nozzles, upstream wall 24 contour profile and mixing chamber combination are 25 configured to encourage working fluid atomisation 26 creating a vapour/droplet mixed flow with local 27 supersonic flow conditions. This encourages the 28 formation of the downstream condensation shock wave, 29 by enhancing local turbulence, pressure gradient and 30 the momentum and heat transfer rate between the 31

1 transport and working fluids by maximising surface contact between the fluids. 2 3 4 The or each nozzle is preferably configured to 5 operate with a particular working fluid, upstream 6 wall contour profile and mixing chamber to provide 7 an optimum nozzle exit pressure. Initial pressure recovery due to transport fluid deceleration, 8 9 coupled with the downstream pressure drop due to 10 condensation, is used to ensure the nozzle expansion 11 ratio is adjusted to enhance atomisation of the 12 working fluid and momentum transfer. 13 14 The exit velocity from the or each nozzle may be 15 controlled by varying the transport fluid supply 16 pressure, the expansion ratio of the nozzle and the 17 condensation profile in the immediate region of the 18 mixing chamber. The nozzle exit velocities may be controlled to enhance Momentum Flux Ratios M in the 19 20 immediate region of the mixing chamber, where M is 21 defined by the equation $M \equiv \frac{\left(\rho_s \times U_s^2\right)}{\left(\rho_s \times U_s^2\right)}$ 22 23 24 where ρ = Fluid density 25 U = Fluid velocity26 Subscript s represents transport fluid 27 Subscript f represents working fluid 28 29 In the present invention it has been found that an 30 optimum Momentum Flux Ratio M for the or each nozzle 31 lies in the range $2 \le M \le 70$. For example, when using

steam as the transport fluid, with a working fluid 1 with a high water content, M for the or each nozzle 2 lies in the range $5 \le M \le 40$. 3 4 The or each nozzle is configured to provide the 5 desired combination of axial, radial and tangential 6 velocity components. It is a combination of axial, 7 radial and tangential components which influence the 8 primary turbulent break-up (atomisation) of the working fluid flow and the pressure gradient. 10 11 The interaction between the transport fluid and the 12 working fluid, leading to the atomisation of the 13 working fluid, is enhanced by flow instability. 14 Instability enhances the droplet stripping from the 15 contact surface of the core flow of the working 16 A turbulent dissipation layer between the 17 transport and working fluids is both fluidically and 18 mechanically (geometry) encouraged ensuring rapid fluid core dissipation. The pseudo-vena contracta 20 is a resultant aspect of this droplet atomisation 21 22 region. 23 The internal walls of the flow passage upstream of 24 the or each nozzle may be contoured to provide a 25 combination of axial, radial and tangential velocity 26 components of the outer surface of the working fluid 27 core when it comes into contact with the transport 28 fluid. It is a combination of these velocity 29 components which inter alia influence the primary 30 turbulent break-up (atomisation) of the working 31

1 fluid and the pressure gradient when it comes into 2 contact with the transport fluid. 3 4 Under optimum operating conditions the 5 disintegration or atomisation of the working fluid core is extremely rapid. The disintegration across 6 the whole bore will typically take place in the 7 mixing chamber within, but not limited to, a 8 9 distance approximately equivalent to 0.66D 10 downstream of the nozzle exit. Under different nonoptimised operating conditions disintegration across 11 12 the whole bore of the mixing chamber, may still 13 occur within, but not limited to, a distance 14 equivalent to 1.5D downstream of the nozzle exit, where D is the nominal diameter of the bore through 15 16 the centre of the fluid mover. 17 18 Recirculation occurs in the flow. recirculation is particularly dominant where 19 20 tangential velocity components of the transport 21 fluid are present. The radial pressure gradients created within the mixing chamber are responsible 22 23 for this flow phenomenon which encourages complete 24 and rapid flow dispersion characteristics across the 25 bore. 26 27 This effect is also created when the pseudo-vena 28 contracta is partially established, i.e. vapour-29 droplet flow is dominant along the mixing chamber 30 boundary. The localised pressure gradient draws 31 flow outwards, causing a region downstream of the 32 transport fluid nozzle exit, typically between 1

diameter and 2 diameters downstream, where the axial 1 flow component of the working fluid stagnates and may even reverse briefly on the centre-line, i.e. 3 the centre of the flow region. 4 5 Recirculation has particular benefits in some 6 applications such as emulsification. 7 8 A series of nozzles with respective mixing chamber 9 sections associated therewith may be provided 10 longitudinally of the passage and in this instance 11 the nozzles may have different angular orientations, 12 for example decreasing from the first nozzle in a 13 downstream direction. Each nozzle may have a 14 different function from the other or others, for 15 example pumping, mixing, disintegrating or 16 emulsifying, and may be selectively brought into 17 operation in practice. Each nozzle may be 18 configured to give the desired effects upon the working fluid. Further, in a multi-nozzle system by 20 the introduction of the transport fluid, for example 21 steam, phased heating may be achieved. This 22 approach may be desirable to provide a gradual 23 heating of the working fluid, enhanced atomisation, 24 pressure gradient profiling or a combinatory effect, 25 such as enhanced emulsification. 26 27 In addition the internal walls of the flow passage 28 immediately upstream of the or each nozzle exit may 29 be contoured to provide different degrees of 30 turbulence to the working fluid prior to its 31

interaction with the transport fluid issuing from 1 2 the or each nozzle. The mixing chamber geometry is determined by the 4 desired and projected output performance and to 5 6 match the designed transport fluid conditions and nozzle geometry. In this respect it will be 8 appreciated that there is a combinatory effect as 9 between the various geometric features and their effect on performance, namely there is interaction 10 between the various design and performance 11 12 parameters having due regard to the defined function 13 of the fluid mover. 14 15 According to a fourth aspect of the present 16 invention a method of processing a working fluid 17 includes 18 presenting a fluid mover to the working fluid, the fluid mover having a straight-through passage of 19 20 substantially constant cross section, 21 applying a substantially circumscribing stream 22 of a transport fluid to the passage through an 23 annular nozzle, 24 atomising the working fluid to form a dispersed 25 vapour and droplet flow regime with locally 26 supersonic flow conditions, 27 generating a supersonic condensation shock wave 28 within the passage downstream of the nozzle by 29 condensation of the transport fluid, the position of 30 the condensation shock wave remaining substantially 31 constant under equilibrium flow,

inducing flow of the working fluid through the 1 passage from an inlet to an outlet thereof, and 2 varying at least one of a group of parameters 3 to change the position of the condensation shock 4 wave, the group of parameters including the inlet 5 temperature of the working fluid, the flow rate of 6 the working fluid, the inlet pressure of the working fluid, the outlet pressure of the working fluid, the 8 flow rate of a fluid additive added to the working 9 fluid, the inlet pressure of a fluid additive added 10 to the working fluid, the outlet pressure of a fluid 11 additive added to the working fluid, the temperature 12 of a fluid additive added to the working fluid, the 13 angle of entry of the transport fluid to the 14 passage, the inlet temperature of the transport 15 fluid, the flow rate of the transport fluid, the 16 inlet pressure of the transport fluid, the internal 17 dimensions of the passage downstream of the nozzle, 18 and the internal dimensions of the passage upstream 19 of the nozzle. 20 21 The term straight-through when used to describe a 22 passage emcompasses any passage having a clear flow 23 path therethrough, including curved passages. 24 25 The fluid additive may be gaseous or liquid. 26 fluid additive is not an essential element of the 27 invention, but in certain circumstances may be 28 beneficial. The fluid additive may comprise a 29 powder in dry form or suspended in a fluid. 30

The parameter varying step may include switching 1 between a plurality of transport fluids or between a 3 plurality of fluid additives. 4 5 The improvements of the present invention may be employed to the fluid mover of the aforementioned 6 patent, and enhance its use in a variety of applications as disclosed in the aforementioned 8 9 patent. These applications range from use as a fluid processor, including pumping, mixing, heating, 10 homogenising etc, to marine propulsion, where the 11 12 mover is submersed within a body of fluid, namely 13 the sea or lake or other body of water. In its 14 application to fluid processing a variety of working 15 fluids may be processed and may include liquids, 16 liquids with solids in suspension, slurries, sludges 17 and the like. It is an advantage of the straight-18 through passage of the mover that it can accommodate 19 material that might find its way into the passage. 20 21 The fluid mover of the present invention may also be 22 used for enhanced mixing, dispersion or hydration 23 and again the combination of the shearing mechanism, 24 droplet formation and presence of the condensation 25 shock wave provides the mechanism for achieving the 26 desired result. In this connection the fluid mover 27 may be used for mixing one or more fluids, one or 28 more fluids and solids in particulate form, for 29 example powders. The fluids may be in liquid or 30 gaseous form. It has been found that the use of the 31 present invention when mixing liquid with a powder 32 of particulate form results in a homogeneous

mixture, even when the powder is of material which is difficult to wet, for example Gum Tragacanth 2 which is a thickening agent. 3 4 The treatment of the working fluid, for example 5 heating, dosing, mixing, dispersing, emulsifying etc 6 may occur in batch mode using at least one fluid 7 mover or by way in an in-line or continuous 8 configuration using one or more fluid movers as 9 required. 10 11 A further use to which the present invention may be 12 put is that of emulsification which is the formation 13 of a suspension by mixing two or more liquids which 14 are not soluble in each other, namely small droplets 15 of one liquid (inner phase) are suspended in the 16 other liquid(s) (outer phase). Emulsification may 17 be achieved in the absence of surfactant blends, 18 although they may be used if so desired. In addition, due to the straight through nature of the 20 invention, there is no limitation on the particle 21 size that can be handled, allowing particle sizes up 22 to the bore size of the unit to pass through whilst 23 emulsification is taking place. 24 . 25 The fluid mover may also be employed for 26 disintegration, for example in the paper industry 27 for disintegration of paper pulp. A typical example 28 would be in paper recycling, where waste paper or 29 broken pieces are mixed with water and passed 30 through the fluid mover. A combination of the heat 31 addition, the high intensity shearing mechanism, the

low pressure region in the vapour-droplet flow and 1 the condensation shock wave both rapidly hydrates 2 the paper fibres, and macerates and disintegrates 3 the paper pieces into smaller sizes. Disintegration 4 5 down to individual fibres has been achieved in 6 tests. 7 8 The straight through aspect of the invention has the 9 additional benefit of offering very little flow 10 restriction and therefore a negligible pressure 11 drop, when a fluid is moved through it. This is of 12 particular importance in applications where the 13 fluid mover is located in a process pipe work and 14 fluid is pumped through it, such as the case, for 15 example, when the fluid mover of the present 16 invention is turned 'off' by the reduction or 17 stopping of the supply of transport fluid. 18 addition, the straight through passage and clear bore offers no impedance to cleaning 'pigs' or other 20 similar devices which may be employed to clean the 21 pipe work. 22 23 A detailed description of the energy transfer 24 mechanism, focussing on the momentum transfer 25 between the transport fluid and working fluid by an 26 enhanced shearing mechanism is best described with 27 reference to the accompanying drawings. By way of 28 example, eight embodiments of geometrical features 29 that may be employed to enhance this energy transfer mechanism in accordance with the present invention 30 are described below with reference to the 31 32 accompanying drawings in which:

1 Figure 1 is a cross sectional elevation of a fluid 2 mover according to the present invention; 3 Figure 2 is a magnified view of the shearing 4 mechanism shown in Figure 1; 5 Figure 3 is a cross sectional elevation of a first 6 embodiment; 7 Figure 4 is a cross sectional elevation of a second 8 embodiment; 9 Figure 5 is a cross sectional elevation of a third 10 embodiment; 11 Figure 6 is a cross sectional elevation of a fourth 12 13 embodiment; Figure 7 is a cross sectional elevation of a fifth 14 embodiment; 15 Figure 8 is a cross sectional elevation of a sixth 16 17 embodiment; Figure 9 is a cross sectional elevation of a seventh 18 embodiment; 19 Figure 10 is a schematic section through the fluid 20 regime of the fluid mover of the present invention; 21 Figure 11 is a schematic drawing of the fluid mover 22 of the present invention in use; 23 Figure 12 is a schematic drawing showing pressure in 24 the fluid mover of the present invention under three 25 different operating conditions; 26

21

Figure 13 is a schematic drawing showing a section

through the fluid mover of the present invention and

the pressure distribution in the fluid mover under

two different condensation shock wave positions; and

Figures 14a and 14b are partial cross sectional 2 views through an eighth embodiment of the fluid 3 mover of the present invention. 4 Like numerals of reference have been used for like 5 parts throughout the specification. 6 7 Referring to Figure 1 there is shown a fluid mover 8 9 1, comprising a housing 2 defining a passage 3 10 providing an inlet 4 and an outlet 5, the passage 3 11 being of substantially constant circular cross 12 section. 13 14 The housing 2 contains a plenum 8 for the 15 introduction of a transport fluid, the plenum 8 16 being provided with an inlet 10. The distal end of 17 the plenum is tapered on and defines an annular 18 nozzle 16. The nozzle 16 being in flow communication 19 with the plenum 8. The nozzle 16 is so shaped as in 20 use to give supersonic flow. 21 In operation the inlet 4 is connected to a source of 22 23 a process or working fluid. Introduction of the 24 steam into the fluid mover 1 through the inlet 10 25 and plenum 8 causes a jet of steam to issue forth 26 through the nozzle 16. Steam issuing from the 27 nozzle 16 interacts with the working fluid in a 28 section of the passage operating as a mixing chamber 29 (3A). In operation the condensation shock wave 17 30 is created in the mixing chamber (3A). 31

In operation the steam jet issuing from the nozzle 1 occasions induction of the working fluid through the 2 passage 3 which because of its straight through 3 axial path and lack of any constrictions provides a 4 substantially constant dimension bore which presents 5 no obstacle to the flow. At some point determined 6 by the steam and geometric conditions, and the rate 7 of heat and mass transfer, the steam condenses 8 causing a reduction in pressure. The steam 9 condensation begins shortly before the condensation 10 shock wave and increases exponentially, ultimately 11 forming the condensation shock wave 17 itself. 12 13 The low pressure created shortly before and within 14 the initial phase of the condensation shock wave 15 results in a strong fluid induction through the 16 passage 3. The pressure rises rapidly within and 17 after the condensation shock wave. The condensation 18 shock wave therefore represents a distinct pressure 19 boundary/gradient. 20 21 The parametric characteristics of the steam coupled 22 with the geometric features of the nozzle, upstream 23 wall profile and mixing chamber are selected for 24 optimum energy transfer from the steam to the 25 working fluid. The first energy transfer mechanism 26 is momentum and mass transfer which results in 27 atomisation of the working fluid. This energy 28 transfer mechanism is enhanced through turbulence. 29 Figure 1 shows diagrammatically the break-up, or 30 atomisation sequence 18 of the working fluid core. 31

Figure 2 shows a magnified and exaggerated schematic 1 of the shearing and atomisation mechanism 18 of the 2 3 working fluid by the transport fluid. believed that this mechanism can be broken down into 4 three distinct regions, each governed by established 5 6 turbulence mechanisms. The first region 20 experiences the first interaction between the 7 8 transport and working fluid. It is in this region that Kelvin-Helmholtz instabilities in the surface 9 contact layer of the working fluid may start to 10 11 develop. These instabilities grow due to the shear 12 conditions, pressure gradients and velocity 13 fluctuations, leading to Rayleigh-Taylor ligament 14 break-up 24. Second order eddies within the fluid 15 surface waves may reduce in size to the scale of 16 Kolmogorov eddies 22. It is believed that the 17 formation of these eddies, in association with the 18 Rayleigh-Taylor ligament break-up, result in the formation of small droplets 28 of the working fluid. 19 20 21 The droplet formation phases may also result in a 22 localised recirculation zone 26 immediately 23 following the ligament break-up region. 24 recirculation zone may enhance the fluid atomisation 25 further by re-circulating the larger droplets back 26 into the high shear region. This recirculation, a 27 feature of the localised pressure gradient, is 28 controllable via the transport fluid's axial, 29 tangential and radial velocity and pressure 30 components. It is believed that this mechanism 31 enhances inter alia the mixing, emulsifying and 32 pumping capabilities of the fluid mover.

1 The primary break-up mechanism of the working fluid 2 core may therefore be enhanced by creating initial 3 instabilities in the working fluid flow. 4 Deliberately created instabilities in the transport 5 fluid/working fluid interaction layer encourage 6 fluid surface turbulent dissipation resulting in the 7 working fluid core dispersing into a liquid-ligament 8 region, followed by a ligament-droplet region where 9 the ligaments and droplets are still subject to 10 disintegration due to aerodynamic characteristics. 11 12 Referring now to Figure 3 the fluid mover of Figure 13 1 and 2 is provided with a contoured internal wall 14 in the region 19 immediately upstream of the exit of 15 the steam nozzle 16. The internal wall of the flow 16 passage 3 immediately upstream of the nozzle 16 is 17 provided with a tapering wall 30 to provide a 18 diverging profile leading up to the exit of the 19 steam nozzle 16. The diverging wall geometry 20 provides a deceleration of the localised flow, 21 providing disruption to the boundary layer flow, in 22 addition to an adverse pressure gradient, which in 23 turn leads to the generation and propagation of 24 turbulence in this part of the working fluid flow. 25 As this turbulence is created immediately prior to 26 the interaction between the working fluid and the 27 transport fluid, the instabilities initiated in 28 these regions enhance the Kelvin-Helmholtz 29 instabilities and hence ligament and droplet 30 formation as foreshadowed in the foregoing 31 description occurs more rapidly. 32

1 An alternative embodiment is shown in Figure 4. 2 3 Again, the fluid mover of Figure 1 and 2 is provided with a contoured internal wall 19 of the flow 4 5 passage 3 immediately upstream of the nozzle 16. 6 The contoured surface in this embodiment is provided 7 by a diverging wall 30 on the bore surface leading 8 up to the exit of the steam nozzle 16, but the taper 9 is preceded with a step 32. In use, the step 10 results in a sudden increase in the bore diameter 11 prior to the tapered section. The step 'trips' the 12 flow, leading to eddies and turbulent flow in the 13 working fluid within the diverging section, 14 immediately prior to its interaction with the steam 15 issuing from the steam nozzle 16. These eddies enhance the initial wave instabilities which lead to 16 ligament formation and rapid fluid cone dispersion. 17 18 19 The tapered diverging section 30 could be tapered 20 over a range of angles and may be parallel with the 21 walls of the bore. It is even envisaged that the 22 tapered section 30 may be tapered to provide a 23 converging geometry, with the taper reducing to a 24 diameter at its intersection with the steam nozzle 25 16 which is preferably not less than the bore 26 diameter. 27 The embodiment shown in Figure 4 is illustrated with 28 29 the initial step 32 angled at 90° to the axis of the 30 bore 3. As an alternative to this configuration, 31 the angle of the step 32 may display a shallower or 32 greater angle suitable to provide a 'trip' to the

Again, the diverging section 30 could be tapered at different angles and may even be parallel 2 to the walls of the bore 3. Alternatively, the 3 tapered section 30 may be tapered to provide a 4 converging geometry, with the taper reducing to a 5 diameter at its intersection with the steam nozzle 6 16 which is preferably not less than the bore 7 diameter. 8 9 Figures 5 to 8 illustrate examples of alternative 10 contoured profiles. All of these are intended to 11 create turbulence in the working fluid flow 12 immediately prior to the interaction with the 13 transport fluid issuing from the nozzle 16. 14 15 The embodiments illustrated in Figures 5 and 6 16 incorporate single or multiple triangular cross 17 section grooves 34, 36 immediately prior to a 18 tapered or parallel section 30, which is in turn 19 immediately prior to the exit of the steam nozzle 20 16. 21 22 The embodiments illustrated in Figures 7 and 8 23 incorporate single or multiple triangular 38 and/or 24 square 40 cross section grooves a short distance 25 upstream of the exit of the steam nozzle 16. These 26 embodiments are illustrated without a tapering 27 diverging section after the grooves. 28 29 Although Figures 1 to 8 illustrate several 30 combinations of grooves and tapering sections, it is 31 envisaged that any combination of these features, or 32

any other groove cross-sectional shape may be 2 employed. 3 The tapered section 30 and/or the step 32 and/or the 4 5 grooves 34, 36, 38, 40 may be continuous or discontinuous in nature around the bore. For 6 example, a series of tapers and/or grooves and/or steps may be arranged around the circumference of 8 the bore in a segmented or 'saw tooth' arrangement. 10 The nature of the flow regime in the fluid mover of 11 12 the present invention is described in more detail 13 below, with reference to Figure 10. 14 15 The transport fluid, usually steam 80, enters 16 through nozzle 16 at supersonic velocity. Wherever 17 the term stem is used, it is to be understood that 18 the term can also be applied to other transport 19 The working fluid, usually liquid 82, flows fluids. 20 at a subsonic velocity into the inlet 4. At the 21 nozzle 16 there is a subsonic liquid core 84 which 22 is bounded by a generally rough or turbulent conical 23 interface with the steam 80 and the region of 24 dispersion 88. As the steam 80 exits the nozzle 16 25 it exhibits local shock and expansion waves 86 and 26 forms a pseudo vena contracta 90. The accelerated 27 region of dispersion 88 (or dissociation) of the 28 liquid core flows at a locally supersonic velocity 29 into the vapour-droplet region 92, in which the 30 vapour is steam and the droplets are the working fluid. Condensation takes place in the supersonic 31 32 condensation zone 94 and the subsonic condensation

The condensation shock wave 17 is produced zone 96. 1 when the condensation, which initiates in the locally supersonic low density region 94, reaches an 3 . exponential rate. The zone 96 immediately after the 4 condensation shock wave 17 has a considerably higher 5 density and is hence subsonic. The condensation 6 shock wave 17 thus defines the interface between 7 these two densities. 8 9 In the liquid phase 98 beyond the condensation zone 10 96 there are small vapour bubbles. The position of 11 the condensation shock wave is controllable over a 12 distance L by adjustment of one of the plurality of 13 parameters described herein. 14 15 The break-up and dispersion of the primary liquid 16 core produces a droplet vapour region. Any liquid 17 instabilities on the primary liquid cone surface 18 18 are amplified to form 'waves'. These waves are 19 further elongated to form ligaments that undergo 20 Rayleigh-Taylor break-up, resulting in the formation 21 of small droplets 28, separated ligaments 24 and 22 larger droplets. 23 24 . The secondary region 24 is thus characterised by the 25 rapid increase in the effective fluid surface area. 26 These droplets 28, of varying size, are then subject 27 to several aerodynamic and thermal effects which 28 ultimately result in their break up to sizes 29 characteristic with the turbulence levels in this 30 region. This results in the vapour-droplet region 31

30

1 which defines the flow regime within the fluid 2 mover. 3 The thickness of the viscous sub layer, comprising 4 the high speed vapour/gas and the locally entrained 5 liquid in droplet or ligament form, increases 6 7 downstream to ultimately extend across the entire 8 The turbulence within this region arises from bore. shear (velocity gradient) and eddies (large scale to 10 Kolmogorov scale), as the flow is essentially of a vapour-droplet consistency. High levels of shear 11 12 exist in the gas/liquid interface. 13 14 A large amount of energy is transferred in this 15 secondary region 24 as a result of further particle 16 break-up. Mass transfer takes place as the shear forces and thermal discontinuities result in the 17 18 droplets becoming ever smaller. The pressure reduces and droplets are evaporated in order to 20 maintain equilibrium in the flow. Heat transfer 21 takes place as equilibrium conditions are reached, 22 ensuring that liquid vapour phase transitions and the inverse transitions all occur within the mixing 23 24 section of the passage 3. In the secondary region 25 there is a very rapid increase in the void fraction $\alpha = \frac{A_{g}}{A_{Tot}}$ 26 27 28 where α = void fraction 29 A_g = area of gas phase (dispersion cone)

 $A_{Tot} = total$ area of pump flow

30

31

Thus the rapid increase in specific volume as the 1 liquid droplets/ligaments are further dispersed, will obviously result in a larger void fraction. 3 Subsequently as the flow conditions begin to 4 approach a state of equilibrium, and due to the 5 geometry within the mixing chamber, the vapour flow 6 is encouraged to follow a condensation profile 7 towards an aerodynamic and condensation shock wave, 8 which is a region of non-equilibrium and entropy 9 production. 10 11 The condensation shock wave arises from the rapid 12 change from a two-phase fluid mixture to a 13 substantially single phase fluid with complete 14 condensation of the vapour phase. Since there is no 15 unique sonic speed in vapour droplet mixtures, non-16 equilibrium and equilibrium exchanges of momentum, 17 mass and energy can occur. In order to achieve a 18 normal condensation shock wave, the velocity of the 19 vapour mixture within the mixing chamber has to be 20 maintained above a certain value defined as the 21 equilibrium sonic speed. For conditions where the 22 vapour velocity is greater than the frozen sonic 23 speed, or where the velocity of the vapour mixture 24 is between the equilibrium and frozen sonic speed, 25 this results in a dispersed or partially dispersed 26 condensation shock wave. These two asymptotic sonic 27 28 speeds are: 29 a_e = equilibrium shock speed. This is the speed at 30 which every fluid is in its correct equilibrium 31 condition, i.e. vapour is vapour, liquid is liquid 32

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a_f = frozen shock speed. This occurs primarily due
         to a 'lag' effect, so that some fluids are not in
  4
         their correct phase, for example the local
  5
         temperature and pressure dictate that a vapour
         should be turning to liquid, but the phase change
  6
  7
        has not happened.
  9
        af and ae are defined as:
 10
          a_f = \sqrt{\gamma \cdot R_v \cdot T_s}
 11
 12
          a_e = \sqrt{\frac{\chi \cdot \gamma \cdot R_v \cdot T_s}{\gamma \left[1 - \frac{R_v \cdot T_s}{h_s} \left[2 - \frac{c \cdot T_s}{h}\right]\right]}}
 13
14
15
          where
16
18
         Y = Ratio of specific heats (the vapour and the
19
         fluid)
20
         R_v = Gas constant for vapour phase (steam)
21
         T_{\text{s}} = \text{Saturation temperature of mixture (vapour and}
22
         fluid)
23
         Cp = Specific heat
24
         H_{fs} = Latent heat of vapourisation
25
         X = Initial vapour quality
26
         \varepsilon = Vapour fraction (gas/liquid)
27
28
         Subscript v, represents vapour (steam)
29
         Subscript f, represents fluid (e.g. liquid)
30
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Frozen flow arises when the interface transport of 1 mass, momentum and energy between the vapour phase 2 and liquid droplets is frozen completely, i.e. the 3 liquid droplets do not take part in the fluid 4 mechanical processes. 5 6 Equilibrium flow arises when the velocity and 7 temperature of the vapour and liquid are in 8 equilibrium, and the partial pressure due to the 9 vapour is equal to the saturation pressure 10 corresponding to the temperature of the flow. 11 12 The secondary flow regime can better be understood 13 by further subdivision into three sub-regions. 14 15 The first sub-region of the secondary flow regime is 16 the droplet break-up sub-region. Just as in the 17 primary zone, where the liquid core is stripped to 18 form the droplet-vapour zone, with the stripping of the ligaments and droplets on the surface, so in the 20 secondary region there is further break-up or 21 dispersion of these separated ligaments, and also 22 the break-up of droplets whose characteristics are 23 unstable in the turbulent flow regime. The dominant 24 mechanism responsible for the break-up in the 25 secondary region is the acceleration of droplets or 26 momentum transfer due to the slip velocity between 27 vapour and liquid. The injection velocity of the 28 vapour in the present invention is important to this 29 functional aspect of the flow regime. If required, 30 multiple nozzles staggered downstream may be used to 31 encourage this aspect. Other parameters such as 32

1 nozzle angle and mixing chamber geometry can be selected to establish favourable flow conditions. Typical break-up mechanisms in this region are 4 dependant on the local velocity slip conditions and the respective working fluid properties. These are gathered into a dimensionless number referred to as 8 the aerodynamic Weber number defined as: $We = \frac{\rho_{v} \cdot \left(U_{f} - U_{v}\right) \cdot D_{f}}{\sigma}$ 10 11 12 where 13 ρ_v = Density of vapour 14 U = Velocity 15 D_f = Hydraulic diameter of fluid 16 σ_f = Surface tension of fluid 17 Typical break-up mechanisms found in the fluid mover 18 19 of the present invention are vibrational break-up, 20 which can be found with ligaments and droplets whose 21 characteristic length is greater than the stable 22 length; catastrophic break-up, which is especially 23 dominant in the liquid-vapour shear layer where We 24 ≥350; wave crest stripping, which occurs where 25 droplets, due to their size, experience large 26 aerodynamic forces causing ellipsoidal shapes, 27 typically where We ≥300; and short stripping, which 28 is the dominant break-up mechanism where daughter 29 and sattelite droplets have been formed following

the ligament stripping and dispersion, typically

30

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where We≥100.

1 The turbulent motion of the surrounding gas, 2 especially where the Reynold numbers are large (Re > 3 10^4), as is usually the case in the present 4 invention, results in large amounts in local energy 5 dissipation and accompanying droplet break-up. 6 fluctuating dynamic pressures resulting from these 7 turbulent fluctuations are dominant in droplet 8 break-up but very importantly it is this energy that ensures extremely effective dispersion and mixing of 10 the fluids in the flow. 11 12 Turbulent pressure fluctuations result in shear 13 forces capable of rupturing fibres or filaments and 14 dissipating powder lumps or similar solid or semi-15 solid matter. In the primary region energy, mass 16 and momentum transfer takes place through a more 17 distinct boundary, associated with the liquid cone 18 dispersion. In the secondary break-up region this transfer is directly related to the turbulence 20 intensity, closely associated with the turbulent 21 dissipation region in the flow. 22 23 The thermal boundary layer, although similar in 24 characteristic to the turbulent dissipation 25 sublayer, represents the effective boundary where 26 evaporation/condensation and energy transfer occur 27 in either an equilibrium state or 'frozen' state. 28 29 Interfacial transport, which begins within the 30 primary cone dissipation, continues into the 31 secondary vapour-droplet region and is characterised 32

by distinct mechanisms enhanced within the fluid 1 2 mover of the invention through vapour introduction 3 conditions, dependent on pressure and velocity, the 4 physical geometry of the steam nozzles and the 5 mixing chamber geometry. This results in a 6 continuous surface renewal process, which together with the turbulence results in a series of renewed 7 8 eddies of various scales. These eddies create bursts arising from the interface of the liquid 10 vapour and the waves formed on ligaments and 11 droplets which are undergoing further break-up. 12 These bursts have a period which is a function of 13 the interfacial shear velocity. These bursts 14 greatly encourage mixing, heat transport and 15 emulsification (droplet size reduction). 16 17 The second sub-region of the secondary flow regime 18 is the subcooled vapour-droplet region. As the vapour mixture flows through the fluid mover of the 19 20 invention its velocity profile is adjusted through 21 fluidic interaction as well as the static pressure 22 gradient which gradually rises due to general ..23 deceleration of the flow. This controlled diffusion 24 of the supersonic flow, balance of natural fluidic 25 and thermodynamic interactions coupled with discrete 26 geometry results in a vapour-droplet state where 27 sub-cooled droplets exist within a vapour dominant 28 phase. The sub-cooled state of this frozen mixture 29 increases until droplet nucleation, and hence 30 condensation, begins to occur very rapidly. 31 point of maximum sub-cooling (Wilson point) 32 determines the point at which the nucleation rate,

which is closely dependent on sub-cooling because of the available surface area for condensation, begins to occur very rapidly, and reaches near exponential 3 The vapour-droplet region within the fluid 4 mover of the invention thus is able to attain near 5 thermodynamic equilibrium within a very short zone. 6 7 The fluid mover of the invention makes special use 8 of geometric conditions created through both 9 geometry and pseudo geometric conditions to ensure 10 the flow conditions upstream of the critical 11 subcooled state deviate from the thermodynamic 12 equilibrium. This ensures maintenance of the 13 desired vapour-droplet region with its desirable 14 droplet break-up, particle dispersion and heat 15 transfer effects. 16 17 The rapid acceleration of the fluid from the primary 18 fluid cone into the vapour region results in an expansion wave, which similarly represents a 20 thermodynamic discontinuity and allows the vapour 21 droplet region to deviate markedly from equilibrium 22 and enter a 'frozen' flow condition. 23 24 Figure 9 shows an embodiment of the fluid mover of 25 the invention in which the geometry of the passage 3 26 has a mixing chamber 3A with a divergent region 50, 27 a constant diameter region 52 and a re-convergence 28 profile region 54. The constant through bore is 29 maintained, but the embodiment of Fig 9 promotes 30 this expansion and non-equilibrium. This offers 31

1 excellent particle dispersion, and good flow, 2 pressure head and suction conditions. 3 The third sub-region of the secondary flow regime is 4 5 the condensation shock region. As a result of the sub-cooled vapour-droplet flow regime within the 6 fluid mover, the point at which exponential condensation begins to occur defines the 8 condensation shock wave boundary. The mixture 10 conditions upstream of the condensation shock wave 11 determine the nature of the pressure and temperature 12 recovery experienced within the fluid mover. 13 The phase change across the condensation shock wave 14 15 obviously results in heat removal from the vapour 16 phase, although there will be an entropy increase 17 · across the condensation shock wave. The ideal operating conditions in the fluid mover of the 18 invention coincide with the formation of a normal 19 20 condensation shock wave, referred to as being 21 discrete, due to its relatively rapid and hence 22 negligible size measured along the X-axis. 23 The nature of the fluid flow in the fluid mover of 24 25 the present invention may better be understood by 26 reference to Figure 12, which shows the distribution 27 of pressure p in the fluid mover over length x along 28 the axis. Reference is made to the two shock 29 speeds, ae and af, defined earlier. 30

Fig. 12a shows condition A and represents the 1 situation where U_{mixture} > a_{e} , where U_{mixture} is the 2 velocity of the vapour/droplet mixture. 3 4 This results in a normal condensation shock wave, 5 with a fairly rapid rise in pressure across the 6 condensation shock wave. The resulting exit 7 pressure is higher than the local pressure at the 8 steam inlet into the bore of the fluid mover. 9 10 Fig. 12b shows condition B and represents the 11 situation where $a_f > U_{mixture} > a_e$. In this case the 12 mixture velocity is higher than the equilibrium 13 shock speed but less than the frozen shock speed. 14 In this condition the condensation shock wave is 15 fully dispersed resulting in a much more gradual 16 pressure rise across the condensation shock wave. 17 18 Fig. 12c shows condition C and represents the situation where $U_{\text{mixture}} > a_{\text{f}}$. In this condition an 20 'unstable' condition arises, with the steam not 21 fully condensing. This is referred to as a 22 partially dispersed condensation shock wave. This 23 results in the start of the formation of a 24 condensation shock wave (with a reasonably steep 25 pressure gradient), the condensation shock wave 26 formation 'stalling', and then restarting again. 27 However, it has been found that the final resulting 28 exit pressure is often higher than for either 29 Condition A or Condition B. 30

There are several mechanisms for determining the 1 2 state of the flow regime in the fluid mover, and using this information in a control system to 3 provide the flow regime that best meets the demands 4 5 of the application. For example one can measure the 6 temperature at a particular point along the length of the mixing chamber, to determine the existence of a vapour-droplet region. Such a method is non-8 9 intrusive since the mixer wall can be of thin 10 section allowing a rapid response to the change in 11 conditions. Multiple temperature probes spaced 12 downstream of one another can be used to monitor the 13 position of the condensation shock wave, as well as 14 to determine the state of the condensation shock 15 wave profile. 16 As a further example the use of pressure sensors 17 . 18 allows the condensation shock wave position to be determined. 19 20 21 With reference to Figures 13 and 14 there is shown a 22 method of using a series of pressure sensors to 23 detect the position of the condensation shock wave 24 in the mixing chamber. When the condensation shock 25 wave 17 is in the position 17A indicated by Case 1, 26 i.e. in the convergent profile portion 3C of the 27 passage 3, the pressure profile is shown with the 28 reference numeral 101. When the condensation shock 29 wave 17 is in the position 17B indicated by Case 2, 30 i.e. in the uniform profile portion 3B of the 31 passage 3, the pressure profile is shown with the 32 reference numeral 102. Pressure sensors P1, P2 and

P3 in the passage 3 can be used to measure the 1 pressure at three points 103, 104, 105 along the 2 The pressure measurements at these points 3 passage. can be used to determine the position of the 4 condensation shock wave 17. Depending on the flow 5 profile required, one or more parameters, as 6 described hereinbefore, can be changed to alter the flow profile and the position of the condensation 8 shock wave 17. 9 10 Figure 14a shows a typical pressure sensor, although 11 it is to be understood that this is not limiting, 12 and any suitable pressure sensor or measuring device 13 may be used. This method of measuring pressures in 14 the mixing chamber is especially suited for 15 condensation shock wave detection, since the 16 measurement technique only needs to measure a change 17 in pressure rather than being calibrated to measure 18 accurate values. 20 The mixing chamber 3A is sleeved with a thin walled 21 inner sleeve 107 of suitable material, such as 22 stainless steel. A thin layer of oil 108 fills the 23 gap between the sleeve 107 and the inner wall 106 of 24 the mixing chamber 3A. The pressure sensor P1 is 25 located through the wall 106 of the mixing chamber 26 and is in contact with the oil 108. When the 27 pressure inside the mixing chamber 3A changes, the 28 sleeve 107 expands or contracts a small amount, 29 thereby increasing or decreasing the pressure in the 30 oil 108, which is then detected by the pressure 31 sensor P1. 32

1 2 In the embodiment of Figure 14b the sleeve 107 is segmented so that the oil is separated by walls 109 3 4 fixed to the sleeve. This results in separate 5 individual chambers of oil 108A, 108B, each with their own pressure sensor P1, P2. A number of 6 7 separate chambers and pressure sensors may be arranged along the wall 106 of the mixing chamber 8 9 3A. 10 11 The advantage of this instrumentation method is that the sleeve 107 provides a clean inner bore, free of 12 13 any crevices or other features in which working 14 fluid or other transported material can become 15 trapped. This is of particular relevance for use in the food industry. In addition, the pressure sensor 16 17 P1 is free from contamination, suffers no wear or 18 abrasion, and does not become blocked. 19 20 A further possible way of monitoring the 21 condensation shock wave is by the use of acoustic 22 signatures. Due to the density variation in the 23 mixer, even during powder addition, it is possible 24 to determine the 'state' of flow which is an 25 indication of vapour flow, and hence the condition 26 of having a condensation shock wave. The mechanisms 27 for determining the state of the flow regime in the 28 fluid mover may of course be combined. 29 30 Figure 11 shows an embodiment of the fluid mover 1 31 with various control means for controlling the 32 parameters of the flow. The inlet 4 is in fluid

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communication with a working fluid valve 66 which 1 can be used to control the flow rate and/or inlet 2 pressure of the working fluid. A heating means or 3 cooling means (not shown) may be provided upstream 4 or downstream of the valve 66 to control the inlet 5 temperature of the working fluid. The outlet 5 is 6 in fluid communication with an optional working 7 fluid outlet valve 68 which can be used to control 8 the outlet pressure of the working fluid. 9 10 A transport fluid source 62, such as a steam 11 generator, is controllable to provide transport 12 fluid through the transport passage 64 to the plenum 13 The source 62 can be used to control the inlet 14 temperature and/or the flow rate and/or the inlet 15 pressure of the transport fluid. 16 17 The nozzle or nozzles 16 may be mounted for 18 adjustable movement such that a nozzle angle control means (not shown) can be used to control the angle 20 of entry of the transport fluid to the passage. 21 22 The internal dimensions of the passage downstream of 23 the nozzle 16 can be adjusted by means of moveable 24 wall sections 60, which can alter the mixing chamber 25 wall profile between convergent, parallel and 26 divergent at a plurality of sections along the 27 mixing chamber 3A. 28 29 An additive fluid source 70 may be provided to add 30 one or more fluids to the working fluid. 31 additive fluid valve 72 can be used to control the 32

1 flow rate of the additive fluid, including to switch 2 the flow on or off as appropriate. Separate heating 3 means may be provided for the additive fluid, which 4 may be a heated liquid, a gas such as steam or a 5 mixture. The additive may be a powder, and may be introduced through a valve means from a secondary 6 hopper. 8 9 Control means such as a microprocessor may be 10 provided to control some or all of the parameters described above as appropriate. The control means 11 12 can be linked to the condensation monitoring 13 devices, such as the pressure sensors P1, P2, P3 14 which monitor the condensation shock wave, or any 15 other sensor means eg temperature or acoustic 16 sensors. 17 18 The versatility of the fluid mover allows the 19 present invention to be applied in many different 20 applications over a wide range of operating 21 conditions. Furthermore the shape of the fluid 22 mover of the present invention may be of any 23 convenient form suitable for the particular 24 application. Thus the fluid mover of the present 25 invention may be circular, curvilinear or 26 rectilinear, to facilitate matching of the fluid 27 mover to the specific application or size scaling. 28 The enhancements of the present invention may be 29 applied to the fluid mover in any of these forms. 30 31 The fluid mover of the present invention thus has 32 wide applicability in industries of diverse

character ranging from the food industry at one end 1 of the chain to waste disposal at the other end. 2 3 The present invention when applied to the fluid 4 mover of the aforementioned patent affords 5 particularly enhanced emulsification and 6 homogenisation capability. Emulsification is also 7 possible with the deployment of the fluid mover of 8 the present invention on a once-through basis thus 9 obviating the need for multi-stage processing. 10 this context also the mixing of different liquids 11 and/or solids is enhanced by virtue of the improved 12 shearing mechanism which affects the necessary 13 intimacy between the components being brought 14 together as exemplified heretofore. 15 16 The localised turbulence within the working fluid 17 dispersion region provides rapid mixing, dispersion 18 and homogenisation of a range of different fluids and materials, for example powders and oils. 20 21 The heating of fluids and/or solids can be effected 22 by the use of the present invention with the fluid 23 mover by virtue of the use of steam as the transport 24 fluid and of course in this respect the invention 25 has multi-capability in terms of being able to pump, 26 heat, mix and disintegrate etc. 27 28 The fluid mover of the present invention may be 29 utilised, for example, in the essence extraction 30 process such as decaffeination. In this example the 31 fluid mover may be utilised to pump, heat, entrain, 32

hydrate and intimately mix a wide range of aromatic 1 2 materials with a liquid, usually water. 3 The vapour-droplet flow region of the present 4 5 invention provides a particular advantage for the 6 hydration of powders. Even extremely hard-to-wet 7 hydrophilic powders, for example Guar gum, may be 8 entrained and dispersed into a fluid medium within 9 this vapour-droplet region. 10 11 As has been disclosed above, the fluid mover of the 12 present invention possesses a number of advantages 13 in its operational mode and in the various 14 applications to which it is relevant. For example 15 the 'straight-through' nature of the fluid mover 16 having a substantially constant cross section, with 17 the bore diameter never reducing to less than the 18 bore inlet, means that not only will fluids 19 containing solids be easily handled but also any 20 rogue material will be swept through the mover 21 without impedance. The fluid mover of the present 22 invention is tolerant of a wide range of particulate sizes and is thus not limited as are conventional 23 24 ejectors by the restrictive nature of their physical

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convergent sections.

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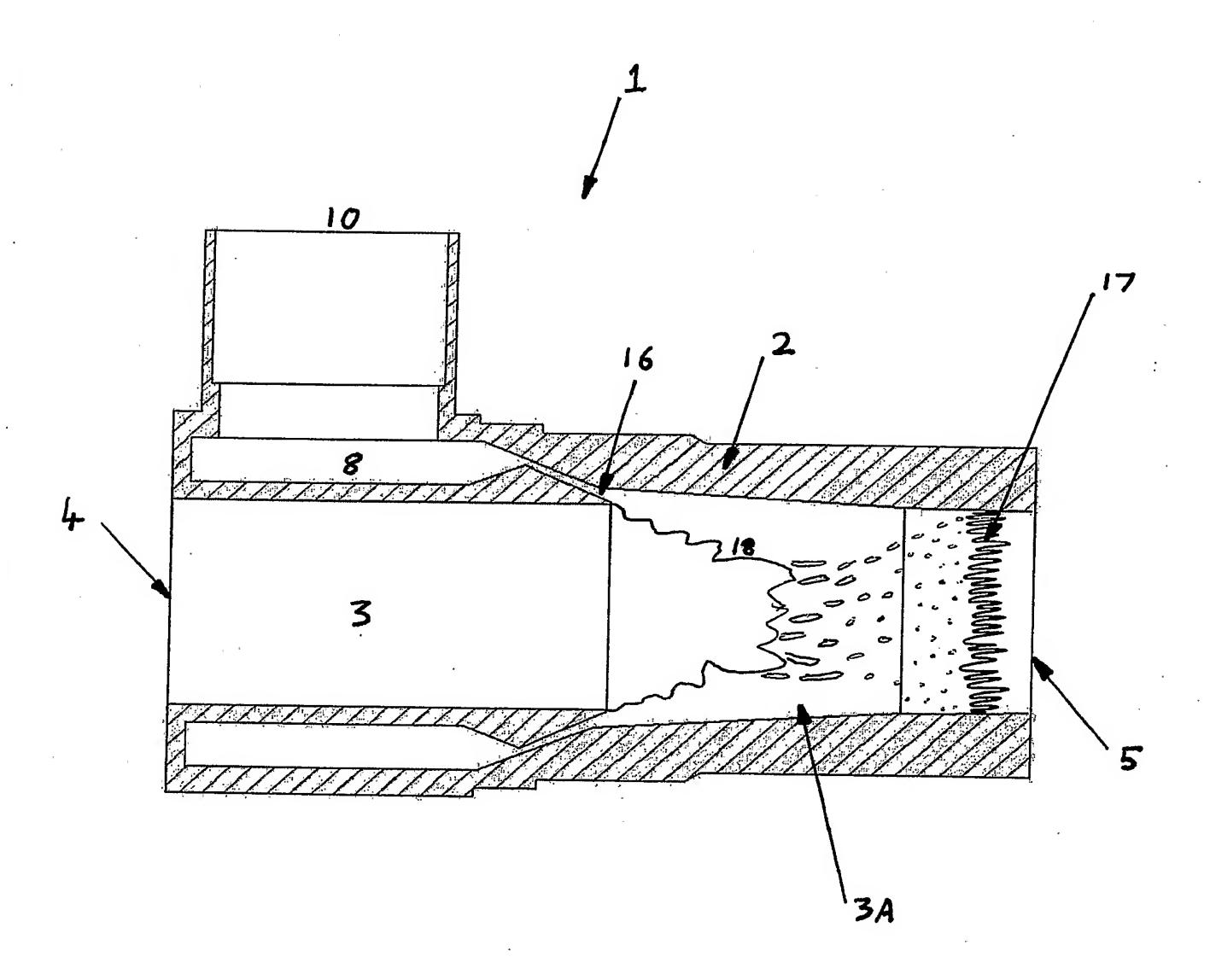


Figure 1

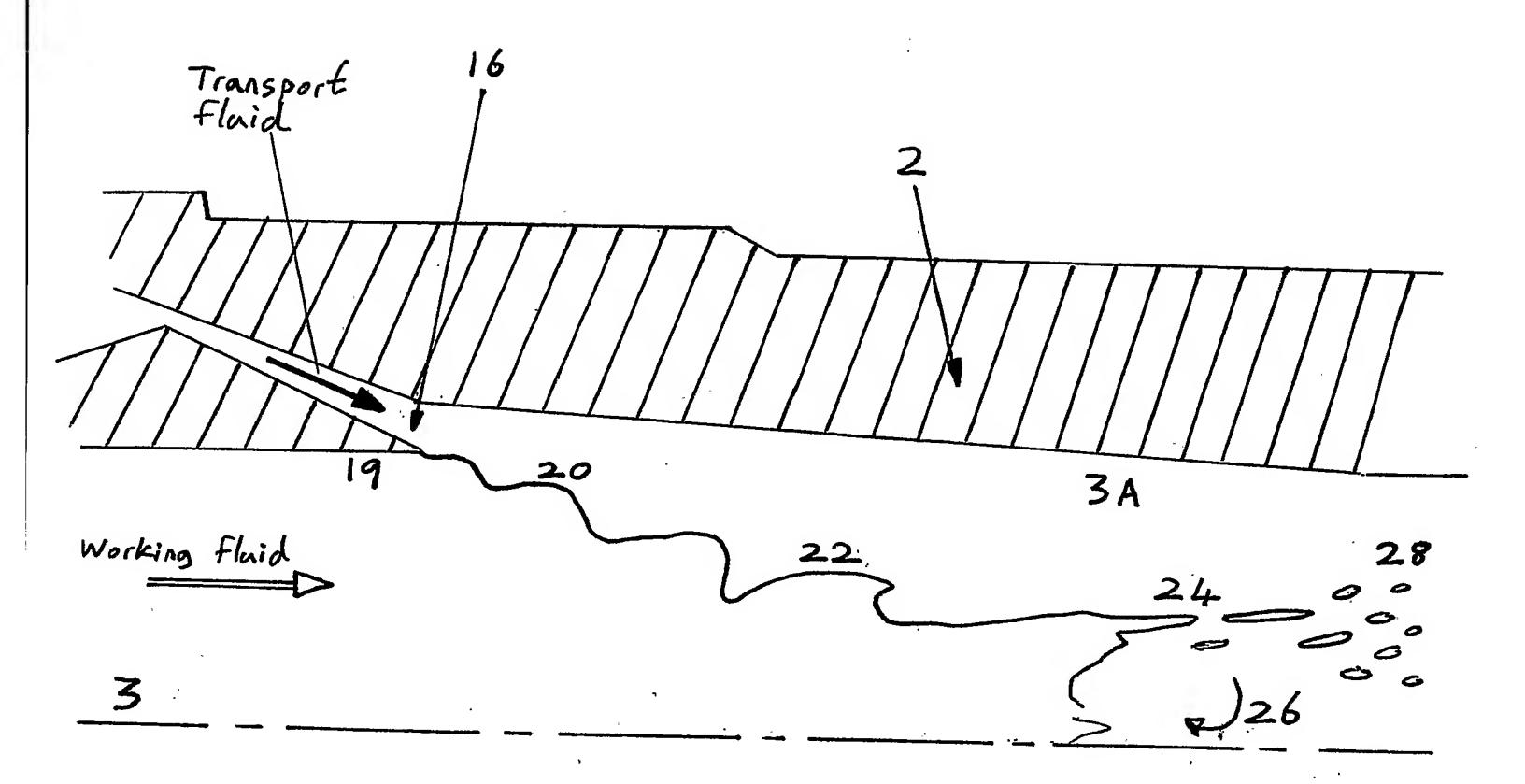


figure 2

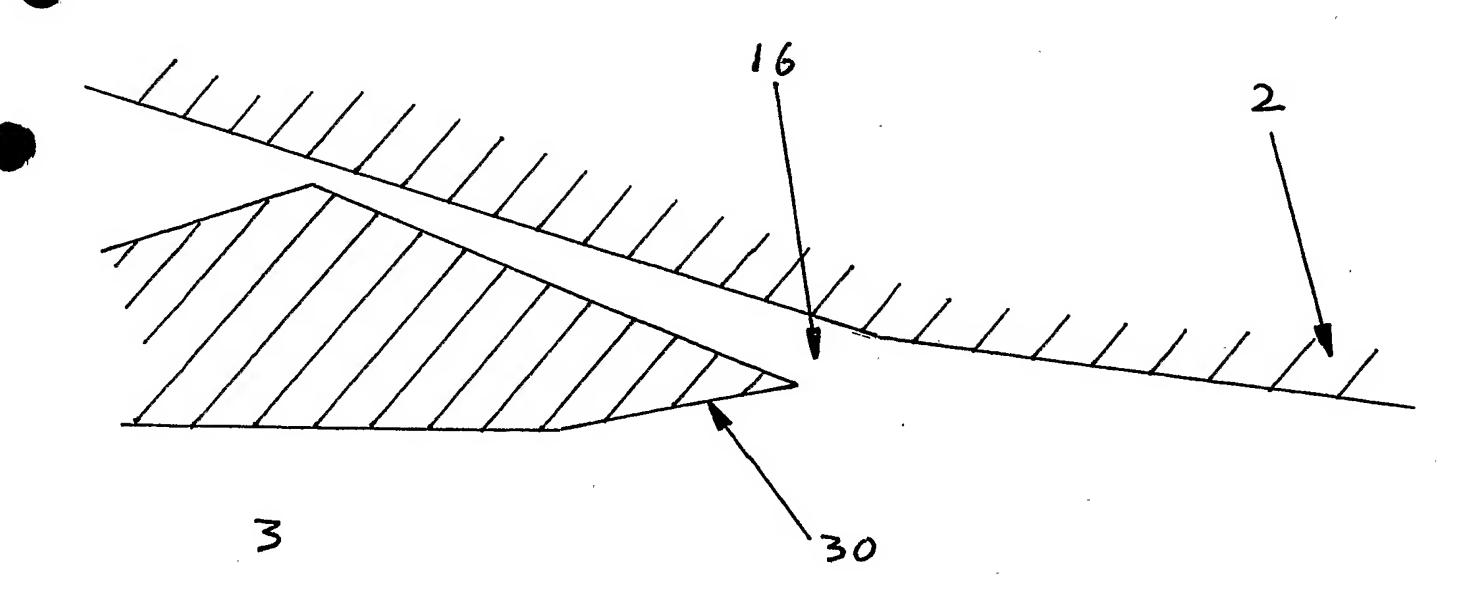


figure 3

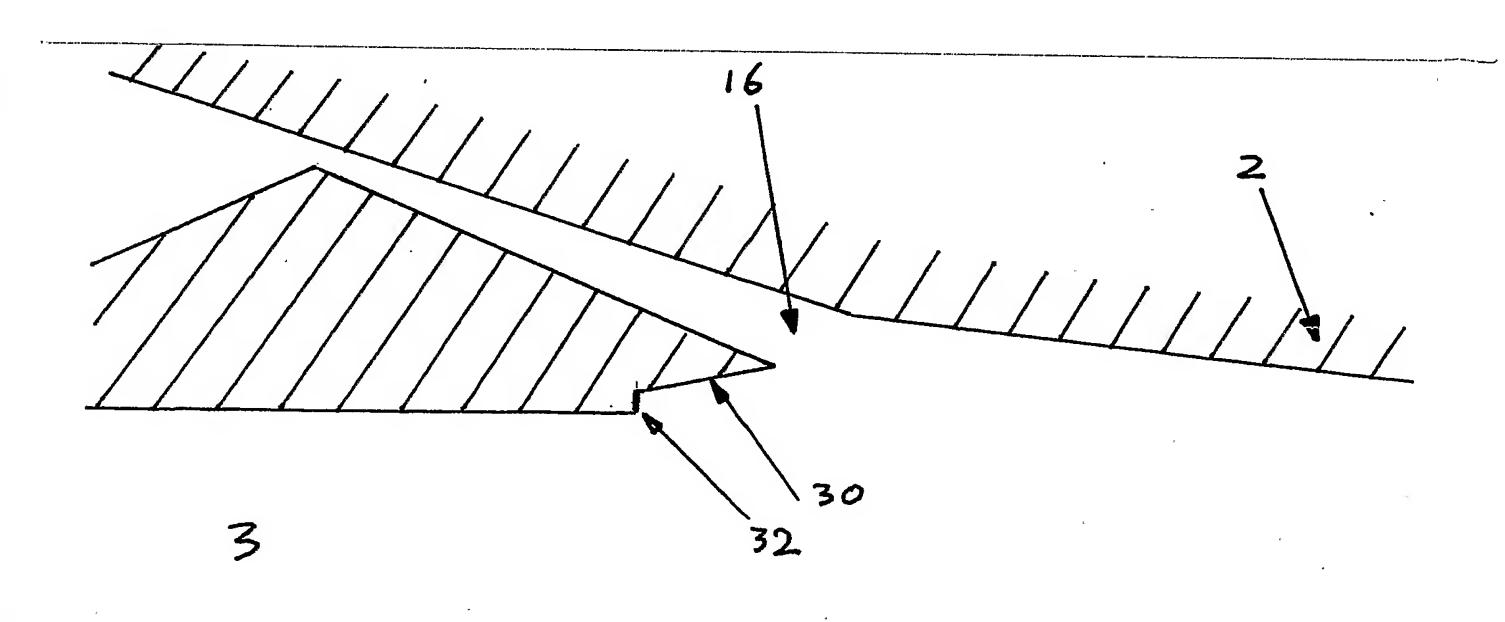


figure 4

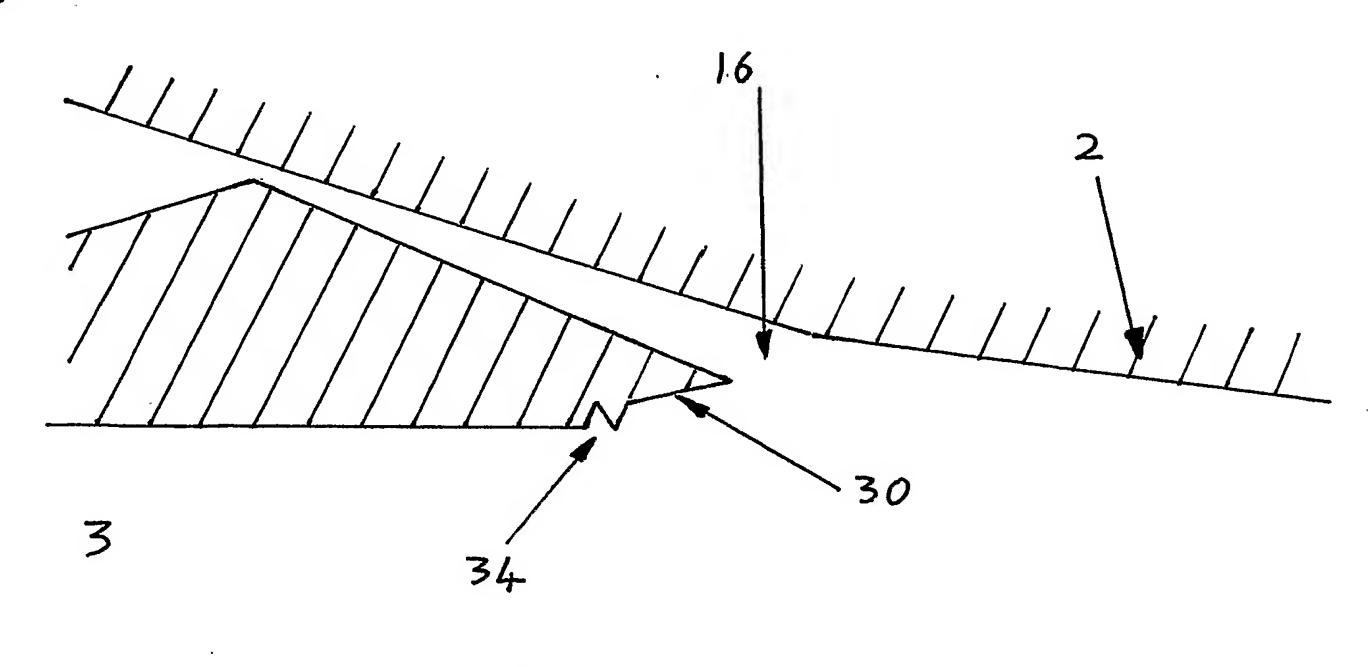


figure 5

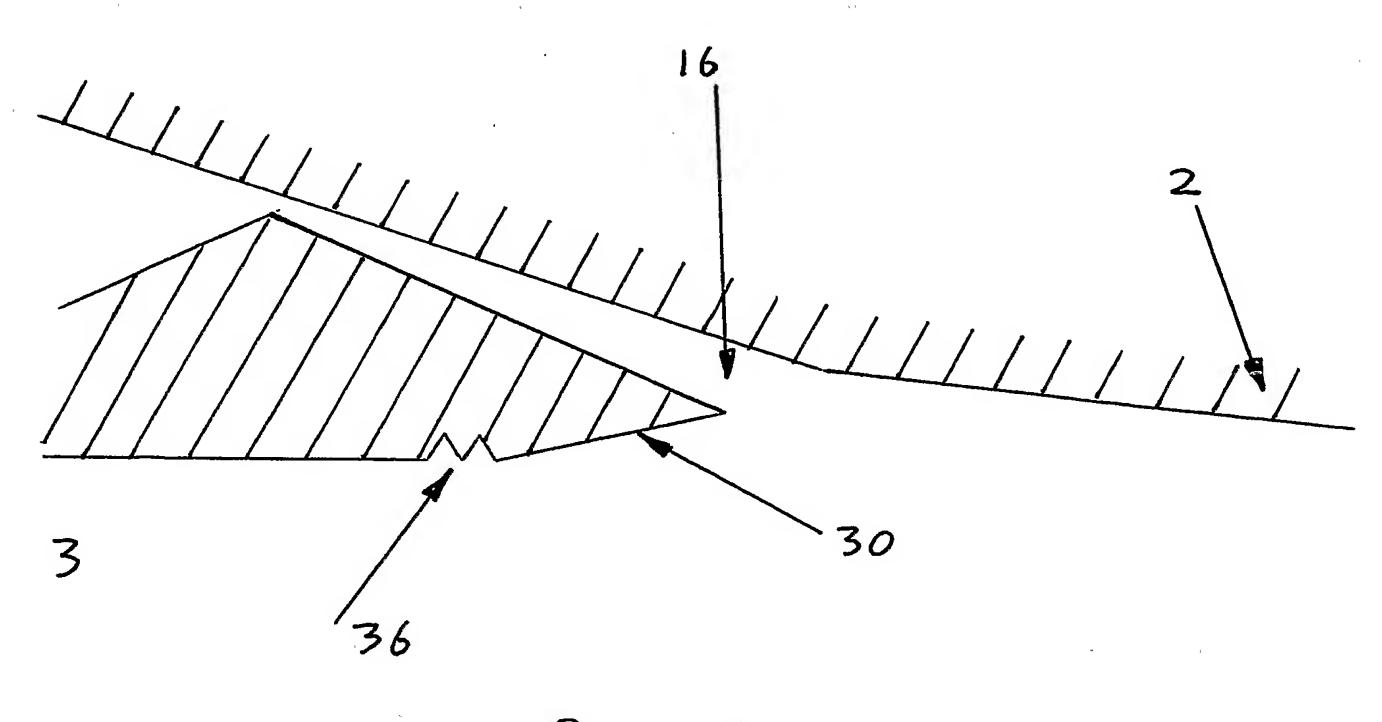


figure 6

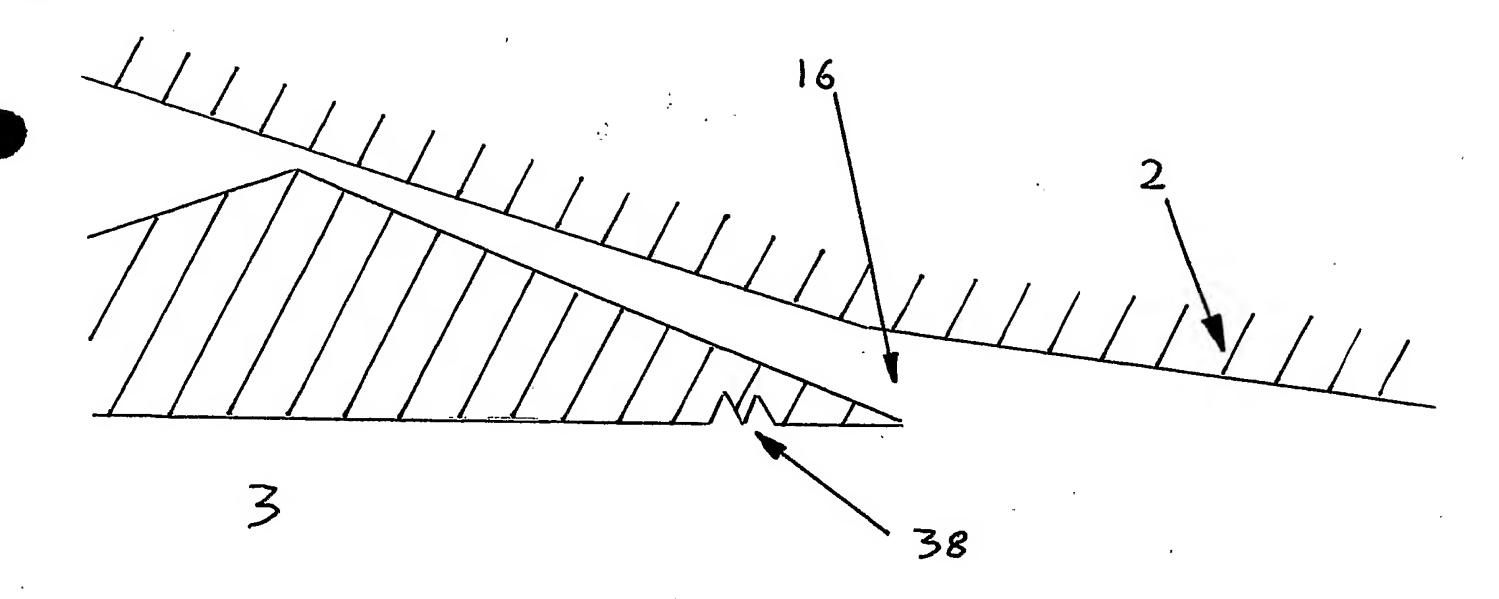


figure 7

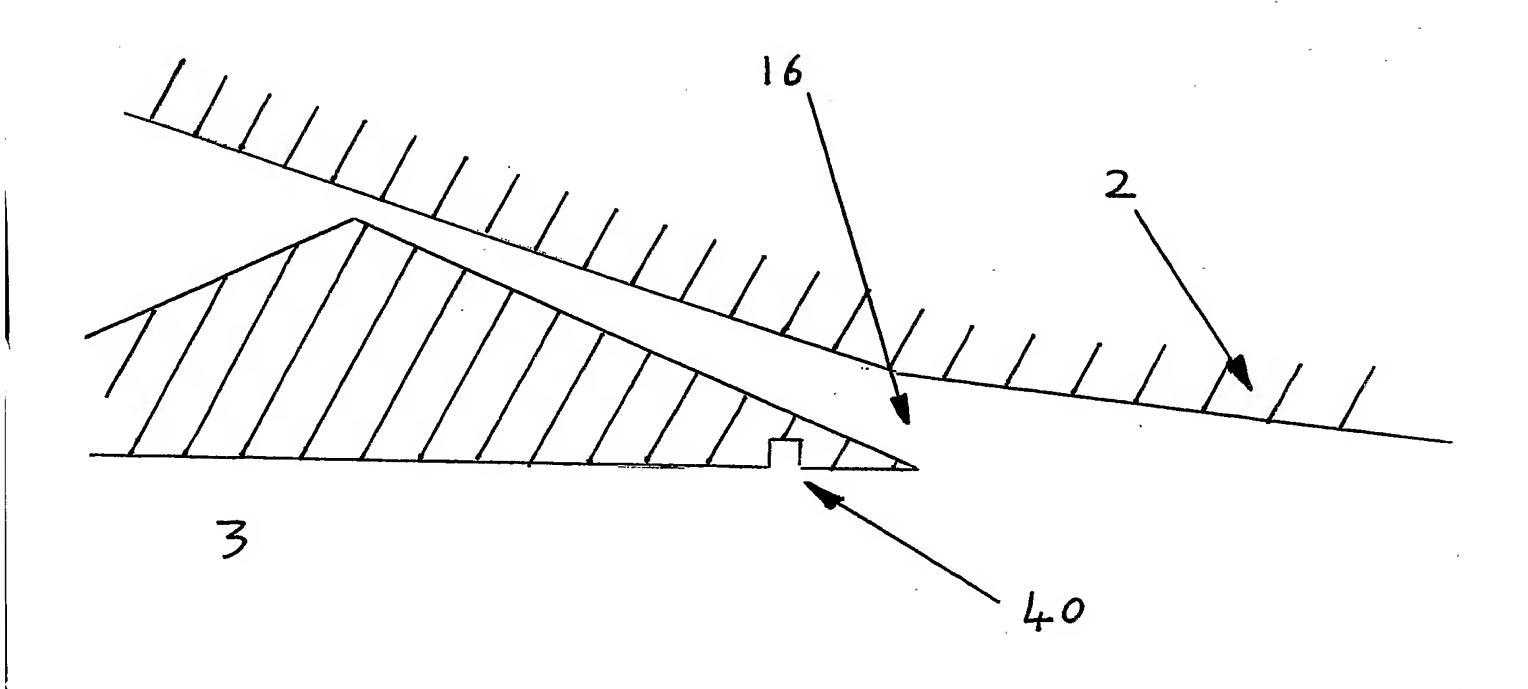


figure 8

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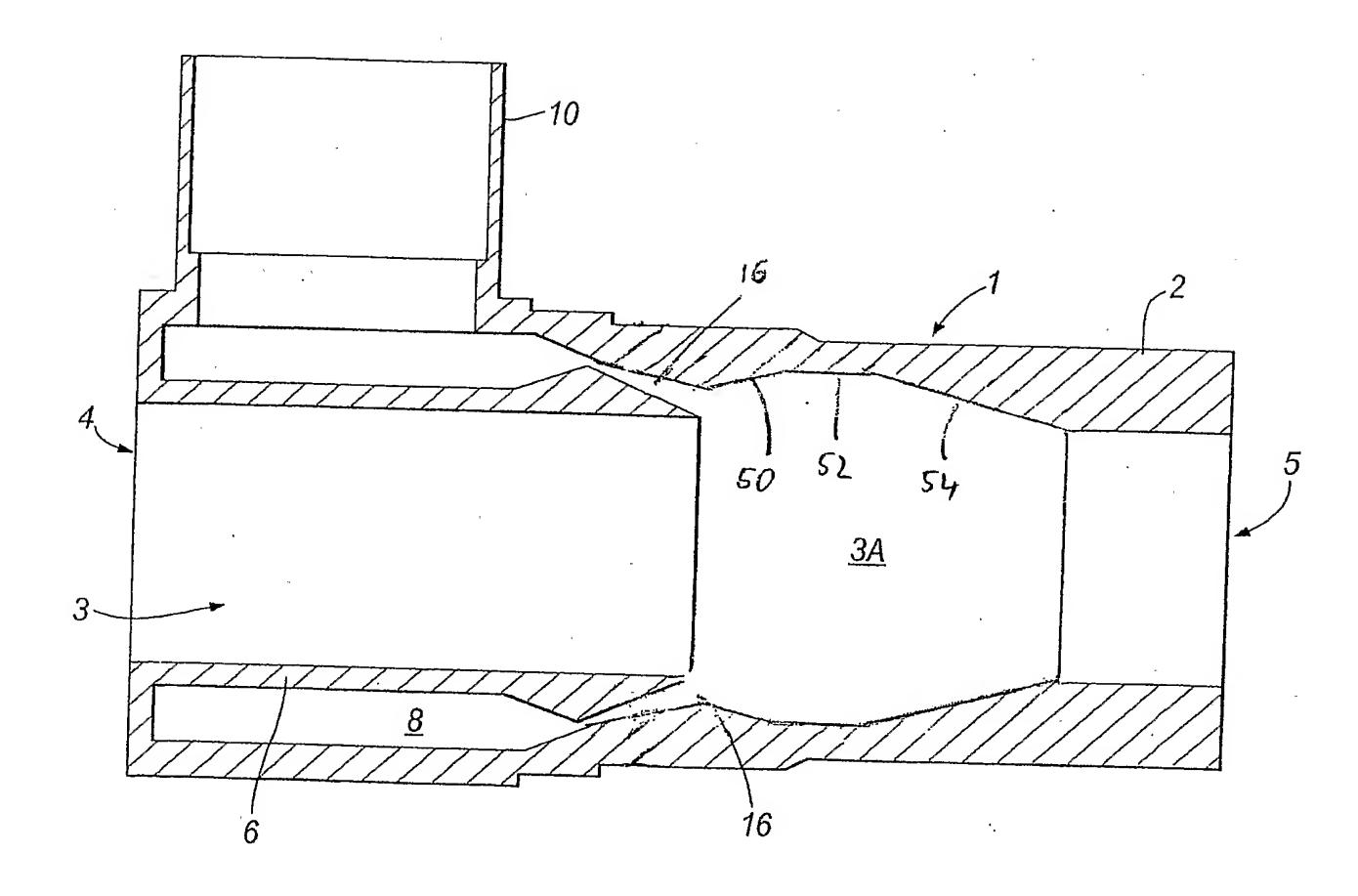
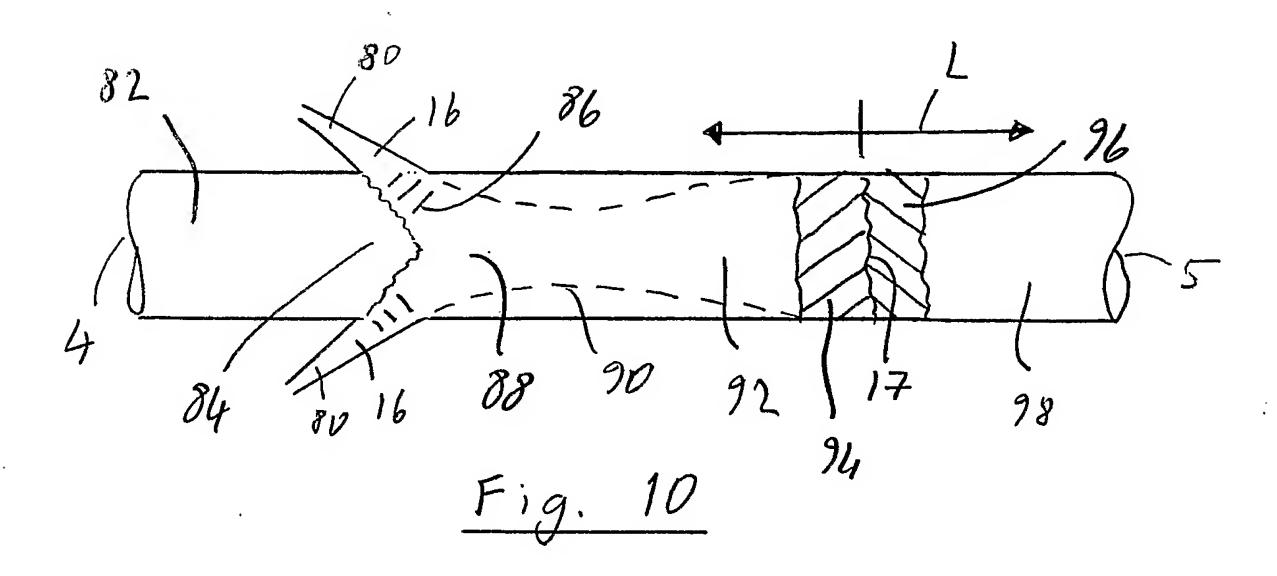
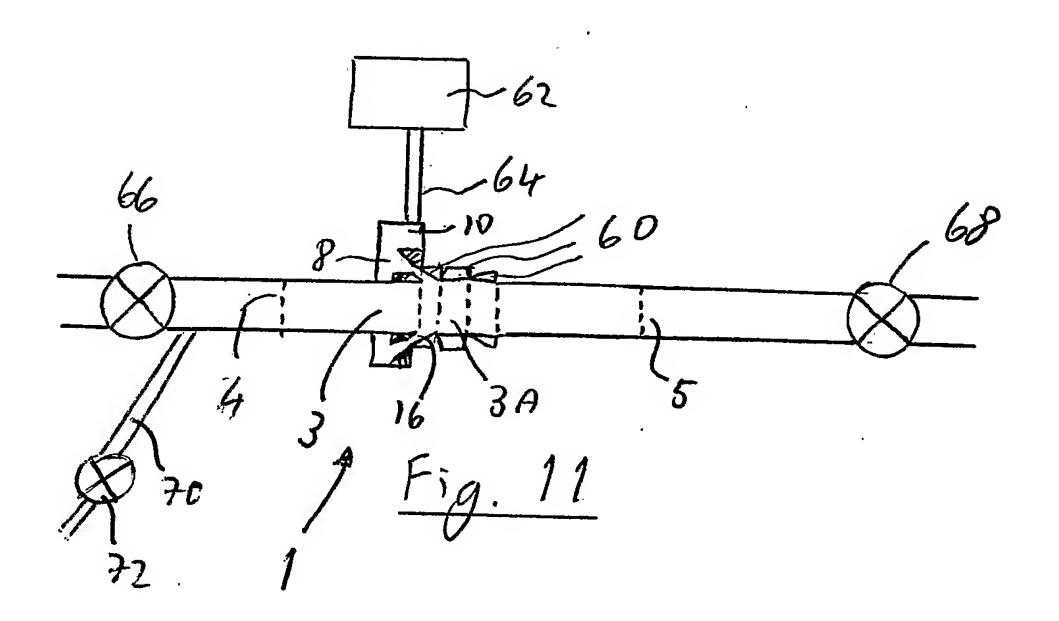


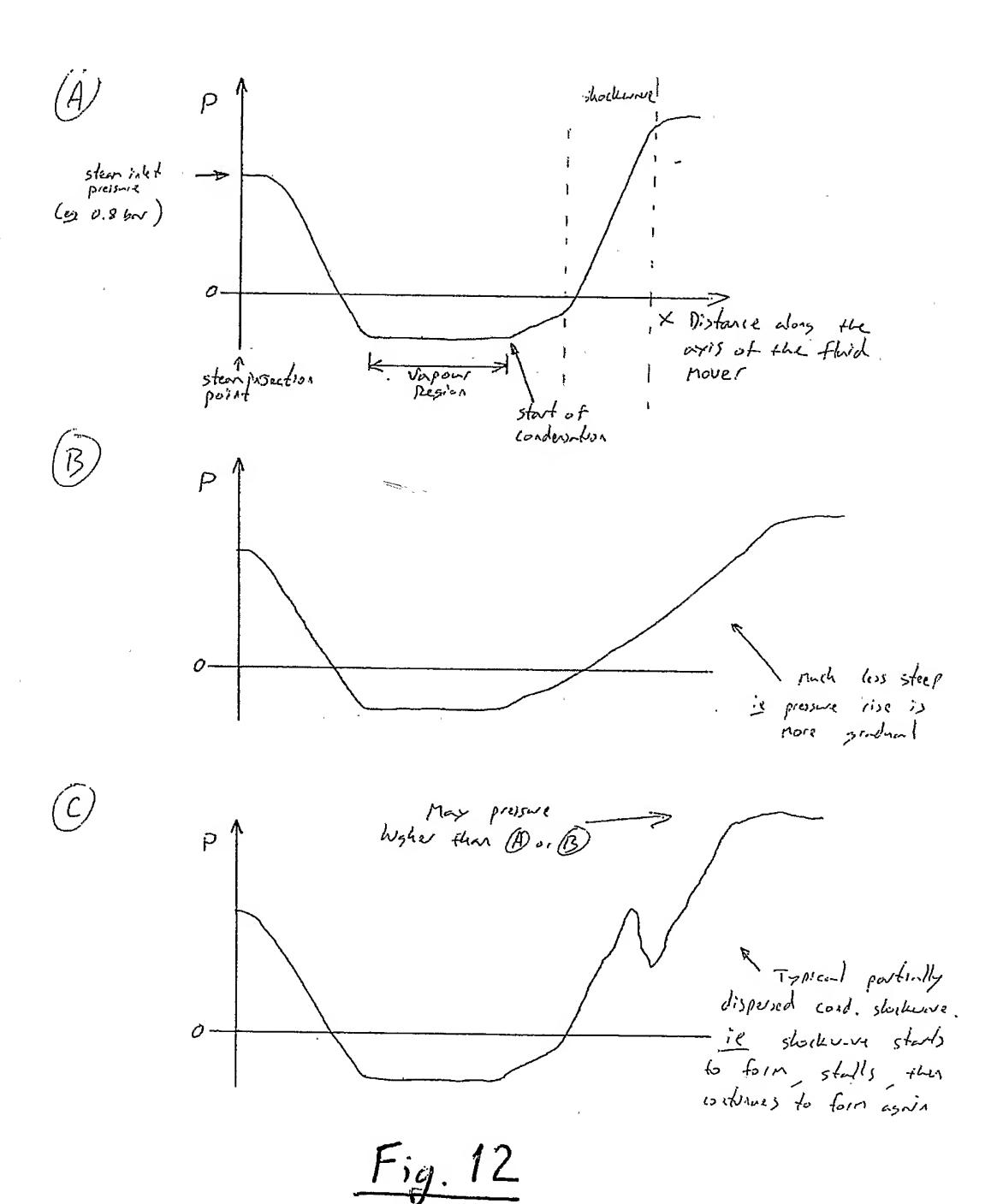
Fig. 9

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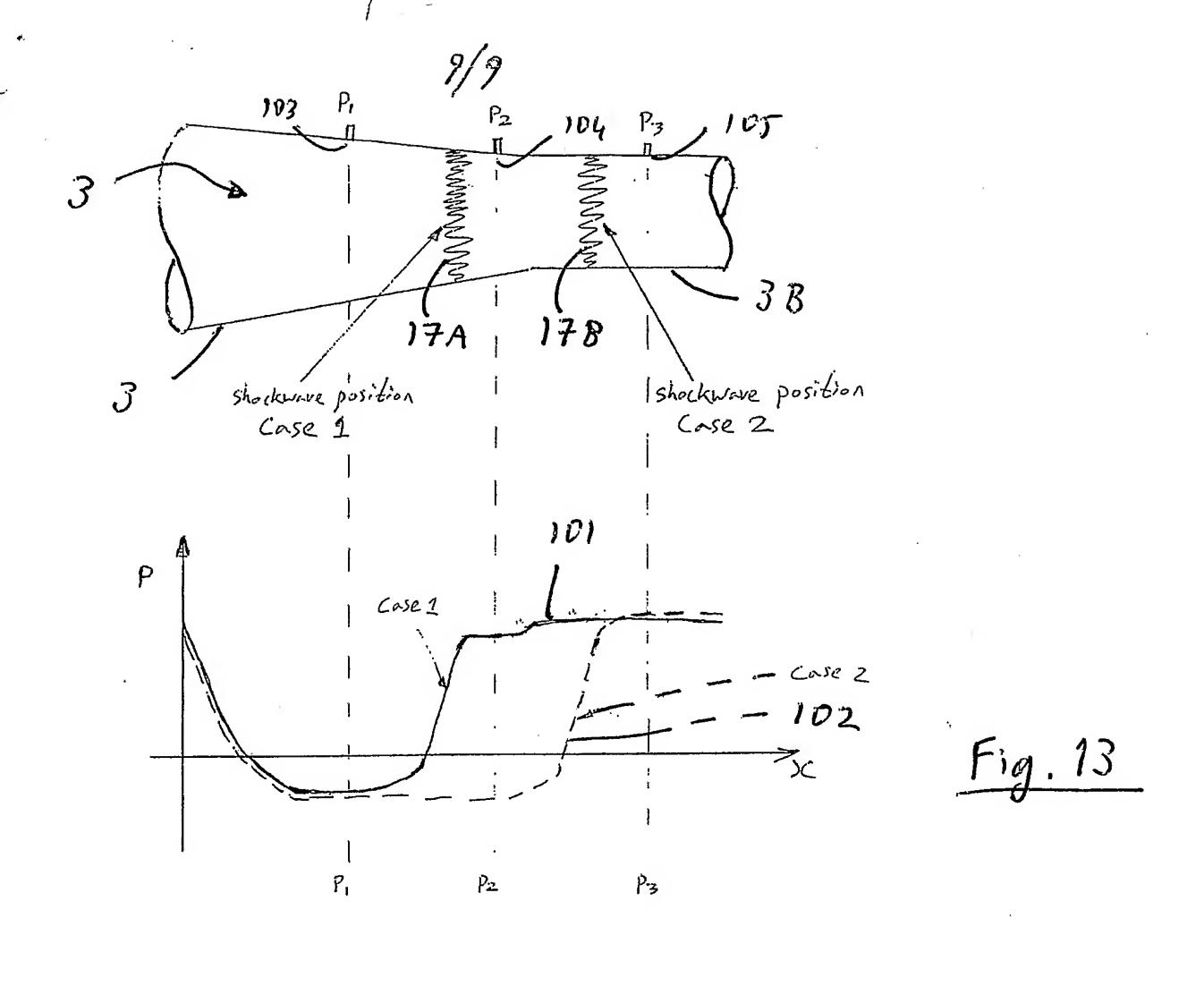


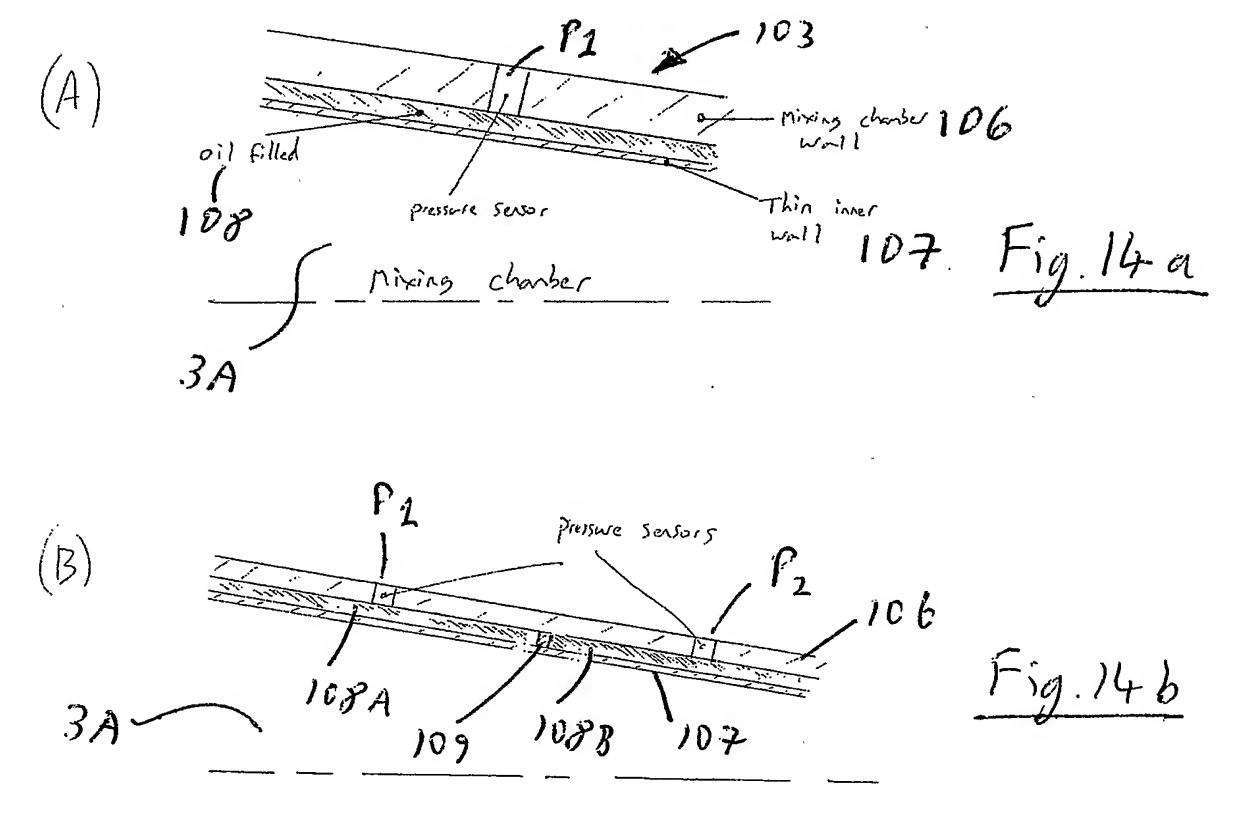


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